demands to reprogram the computer for different memory map configurations several times a day. Command and computer operations were done solely at APL.

By the first month of flight, all data-harvesting and experiment-controlling programs had been exercised. The computer successfully provided the following services:

- 1. Support of attitude stabilization using delayed command function;
- 2. Data storage and processing in conjunction with the attitude, Disturbance Compensation System, Environmental Survey Package, Radioactive Thermal Generator, and thermal systems;
- 3. Broadcast of a valid navigation message resulting in a successful ground navigation experiment;
- 4. Operation of the on-board GMT clock using the two timekeeping systems;
- 5. Exercise of the special Pseudo Random Noise mode on the 160-MHz downlink.

After a month in orbit, a failure in the telemetry system put an end to 90% of the possible telemetry functions. This drastically curtailed the real-time data available during satellite passes and eliminated the usefulness of most of the data-gathering flight programs.

After the failure, the flight computer became the only means of monitoring the Disturbance Compensation System and its fuel consumption rate. The Disturbance Compensation System thruster program was used extensively in this capacity to support a long-term Disturbance Compensation System experiment. Other major computer activities during this period included broadcasting 50 bits/sec navigation data and managing the on-board clock.

A second failure at the end of the second month—this time in the computer injection logic —prevented further access to the computer. Although all experiments using the computer had been run as planned, the early system failures meant a disappointing data yield.

In judgment of the flight software system, all of the requested combinations of activities were accomplished within the on-board constraints. The realization of continuous computer activity ranged between one and two weeks between disruptions. These problems were due largely to the complex ground support operations needed for TRIAD and the varying operational procedures for different experiments.

Failure notwithstanding, the TRIAD flight computer proved to be an extremely useful and versatile tool in a satellite with such diverse experimental and operational requirements.

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WITH THE AUTHORS



T. L. McGovern, co-author of "TRIAD Incremental Phase Shifter," a native of Maryland, received his B.S. degree in physics from The Johns Hopkins University. Prior to employment at APL in 1961, he was employed at Melpar Inc. where he specialized in electronic airborne reconnaissance equipment and military aircraft flight simulators, and in the U.S. Army at Huntsville, Alabama, as an electronics specialist in surfaceto-air guided-missile systems. Mr. McGovern has been with the Space Development Department since his employment with APL and has principally been involved with the digital design of flight systems and exploratory processing systems for ground support. Mr. McGovern is a member of the American Physical Society and The American Association for the Advancement of Science.

L. J. Rueger, co-author of "TRIAD Incremental Phase Shifter," a native of Ohio, received the B. S. degree in engineering physics and M. S. degree in physics from the Ohio State University. A specialist in microwave research, precision frequency measurement techniques, and electronic system engineering, he joined APL after positions at the MIT Radiation Laboratory, Battelle Memorial Institute, and the National Bureau of Standards. Originally a member of the missile Guidance Group, Mr. Rueger subsequently worked on several projects including the development of radar map-matching techniques, the development of electronic countermeasures, and an atomic frequency standards facility. Since 1964,



he has been Supervisor of the Space Electronics Systems Group in the Space Development Department, as well as Project Scientist for the Navy Navigation Satellite System. Mr. Rueger has been Consultant to the Navy on military specifications for shipboard navigation equipments since 1967. He is a member of several professional societies including the American Physical Society and the Institute of Electrical and Electronics Engineers.

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WITH THE AUTHORS (continued)



J. A. Perschy, co-author of "TRI-AD Programmable Computer," was born in Washington, D.C. He worked as field engineer on the UNIVAC I computer at the Bureau of Census while attending The George Washington University School of Engineering. In 1958, after receiving a B.S. degree in electrical engineering from The George Washington University, he joined APL as Associate Engineer. He advanced to Engineer while working on satellite memory systems in 1962, and received an M.S. in physics from the University of Maryland in 1965. A specialist in aerospace computers and real-time systems, Mr. Perschy is a section supervisor in the Space Digital Systems Group. He designed the datahandling system and memory for the 4A, 5A, and OSCAR series of navigation satellites and the general purpose computer for the TIP series of satellites. His earth-bound designs include the computer for a sonar intercept receiver and the digital measurement unit for a Loran receiver-navigator.



B. M. Elder, co-author of "TRI-AD Programmable Computer," was born in Dallas, Texas. He received a B.E.E. degree from the University of Florida and has taken additional work in computer design and information systems engineering. A specialist in digital circuit and system design and development, he was employed at APL as Associate Engineer in 1961 after having been with Melpar, Inc. He advanced to the rank of Engineer in 1964 and since then has been a section supervisor in the Space Digital Systems Group. Mr. Elder had design responsibility for the ANNA satellite memory system, contributed to the design and breadboard evaluation of a satellite memory system for the NAPOR program and the design of the OA navigation satellite memory system. Mr. Elder shared design, testing, and system integration responsibility for a direct memory access interface unit for the Honeywell DDP 516 computer and a Radar Video Processor, and, more recently, for the TRIAD spacecraft computer. H. K. Utterback, co-author of "TRIAD Programmable Computer," was born in Chambersburg, Pa., and received the A.B. degree in mathematics at Gettysburg College. He was employed at APL as Associate Mathematician in the Computing Center in 1960 after having served with the National Security Agency at Fort Meade, Maryland. He left APL in 1962 to go to Con-



trol Data Corporation where he participated in a wide variety of software development including assemblers, macroprocessors, real-time monitors, war games and a personnel information system. Mr. Utterback returned to APL in 1969 as Programmer in the Space Analysis and Computation Group. His activities include the development of programs for predicting the tropospheric effect on electromagnetic range measurements from surface meteorological data, and the development of the ground support system for the TIP series of navigation satellites.